



The University of Texas at Austin

Simultaneous N_2 and CO measurements with broadband nanosecond CARS for graphite ablation in an inductively coupled plasma torch

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<https://pecos.oden.utexas.edu>



Predictive
Engineering &
Computational Science



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**Aerospace Engineering
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Acknowledgements

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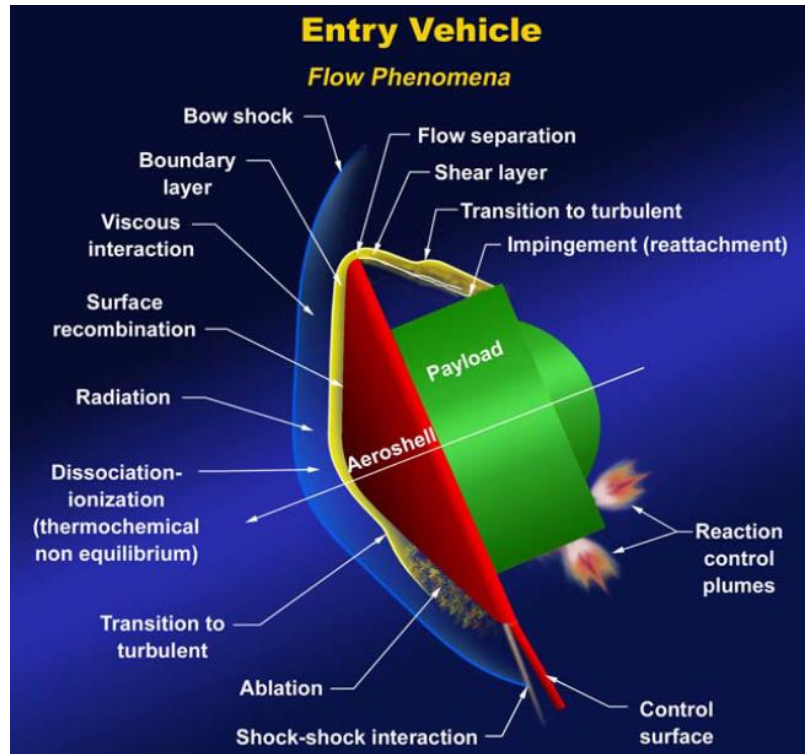
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Development of Thermal Protection Systems (TPS)



Hypersonic atmospheric re-entry: Horvath et al., *NASA Langley Tech. rep.* NASA Langley Research Center (2004)

- TPS are used for
 - spacecraft during atmospheric re-entry.
 - hypersonic aircraft during flight.
- Processes at TPS are complex multi-physics problems.
- Radiative heat load has big impact on TPS design and weight¹.
- Graphite based TPS: CN is a strong radiator.
- CN concentration influenced by reactions forming: N, O, and CO^{2,3}.

[1] Caillault et al., "Radiative heating predictions for Huygens entry". *J. of Geophys. Res.: Planet* (2006)

[2] Park et al., "Chemical-Kinetic Parameters of Hyperbolic Earth Entry". *J. Thermophys. Heat Tr.* (2001).

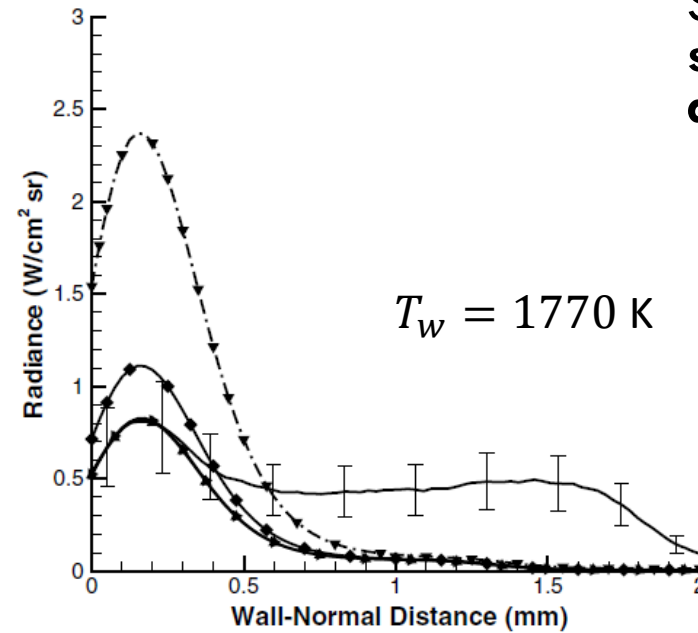
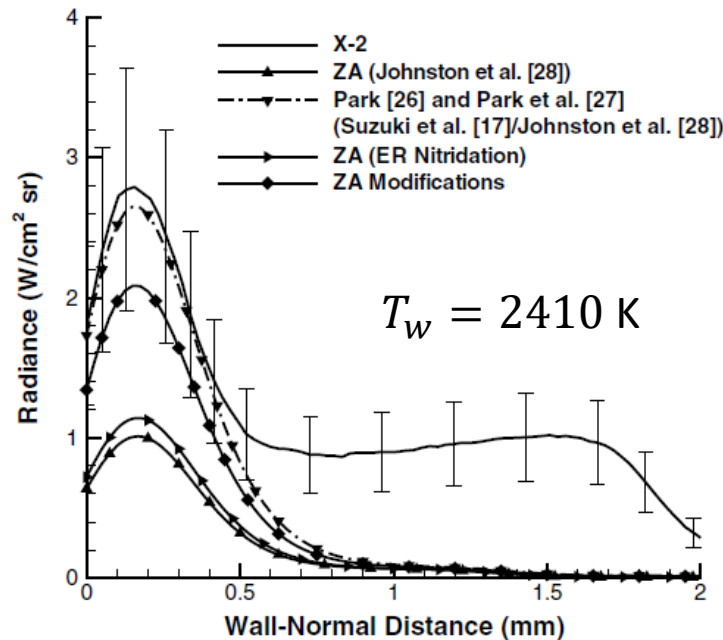
[3] Alba et al., "Development of a nonequilibrium finite-rate ablation model for radiating earth reentry flows". *J. Spacecraft Rocket* (2016)

Challenges in the Development of Thermal Protection Systems

- Different models disagree with each other¹
- Models have disagreed with experiments²
- More recently: new model development³ using molecular beam experiments⁴

- Judging model accuracy requires additional experimental data
- Oxidation processes are of great interest for radiation predictions²

→ Utilize nanosecond multiplex Coherent Anti-Stokes Raman Scattering (CARS) to enable the **spatially resolved** simultaneous probing of **CO and N₂**

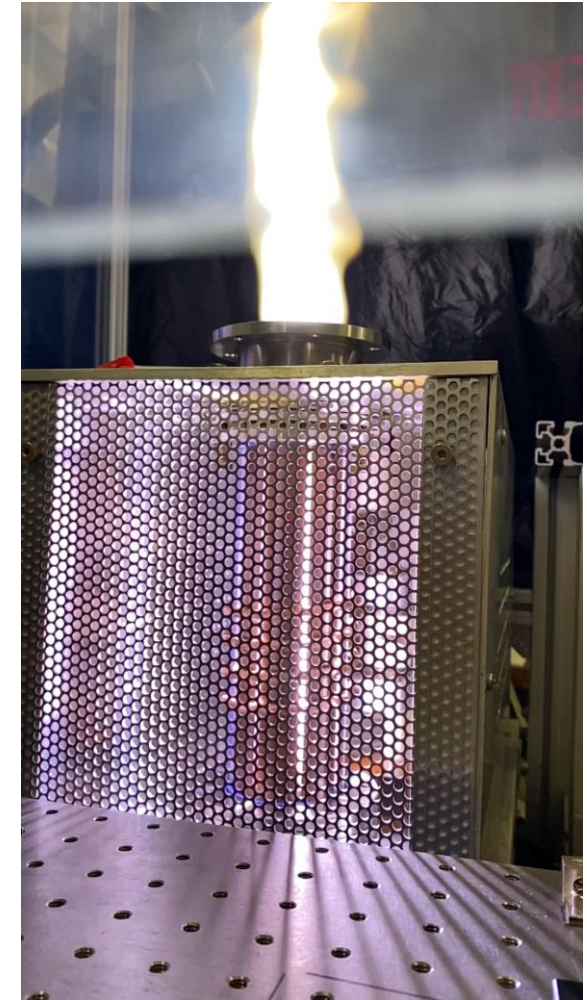
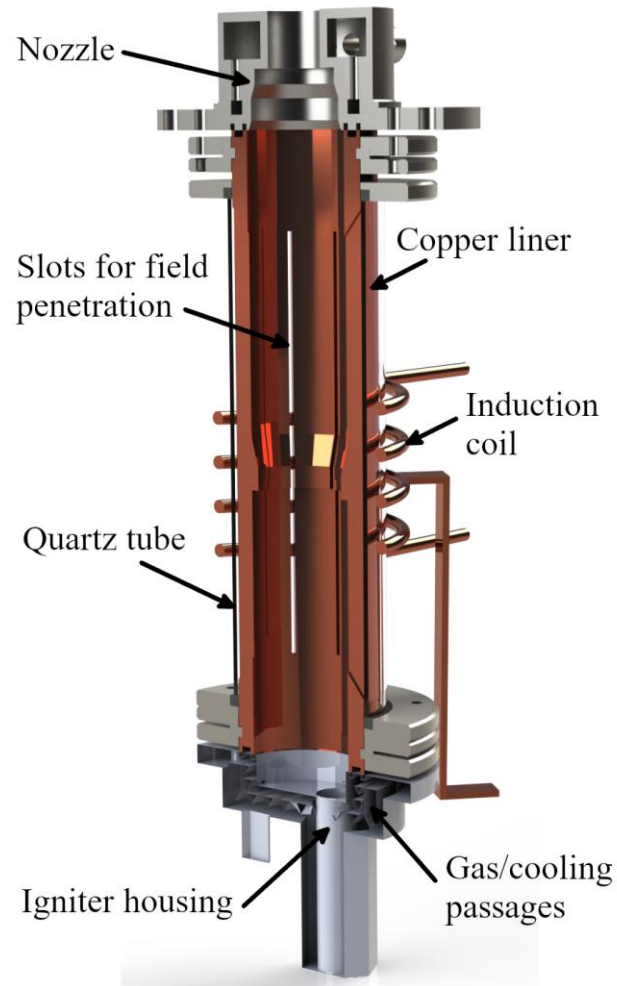


Figures from Alba et al. [2], results for different wall temperatures

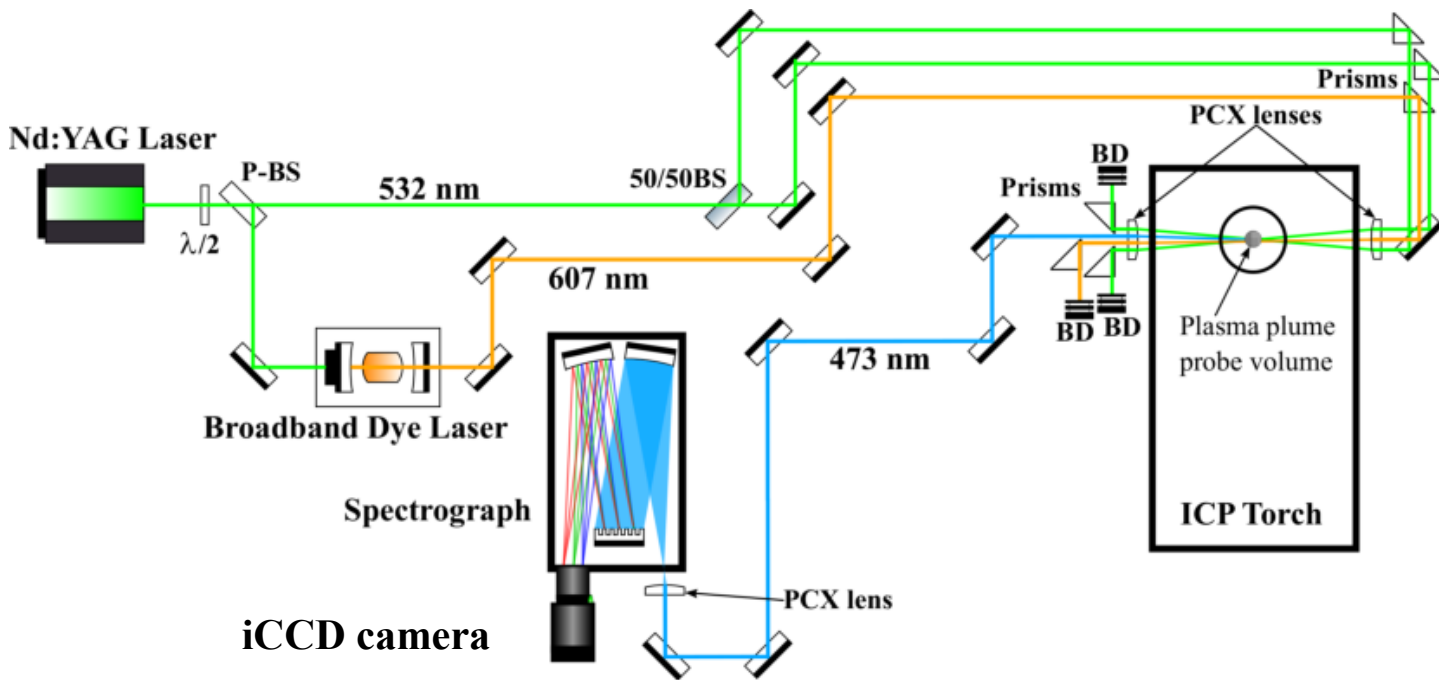
- [1] MacLean et al., "Finite-rate surface chemistry model, II: Coupling to viscous Navier-Stokes code". *42nd AIAA Thermophysics Conference* (2011)
- [2] Alba et al., "Development of a nonequilibrium finite-rate ablation model for radiating earth reentry flows". *J. Spacecraft Rocket* (2016)
- [3] Poovathingal et al., "Finite-rate oxidation model for carbon surfaces from molecular beam experiments". *AIAA J.* (2017)
- [4] Murray et al., "Inelastic and reactive scattering dynamics of hyperthermal O and O₂ on hot vitreous carbon surfaces". *J. Phys. Chem. C* (2015)

Experimental Setup – Inductively Coupled Plasma Torch

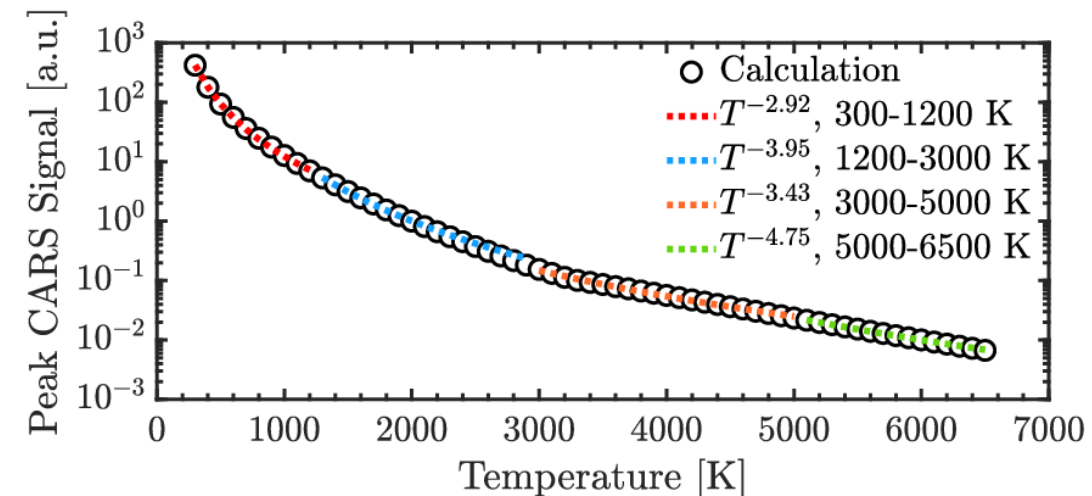
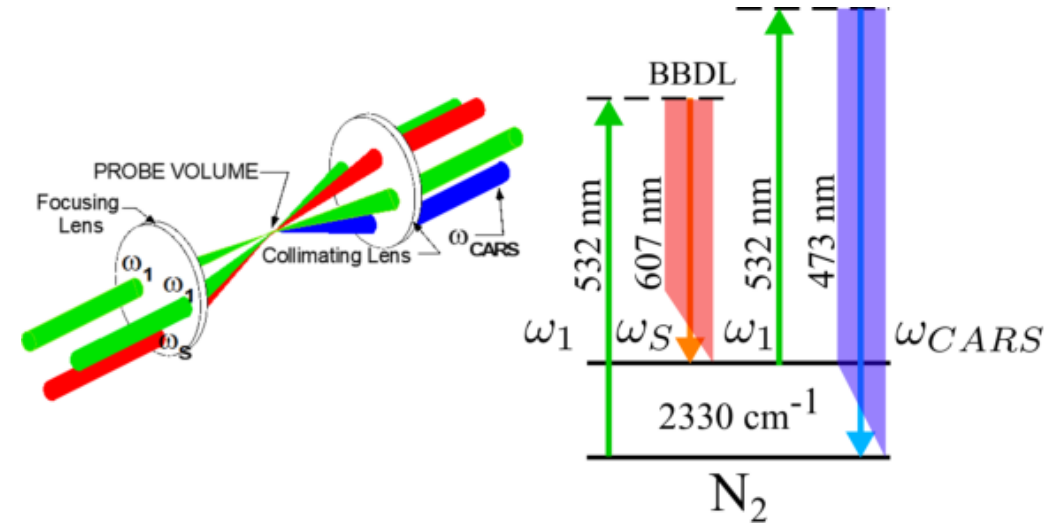
- Atmospheric pressure
- Nozzle diameter 30 mm
- Air plasma
- Exit velocity ~ 15 m/s
- Exit temperature ~ 6000 K @ 5-20 mm from the nozzle exit
- Plume conditions near thermodynamic equilibrium



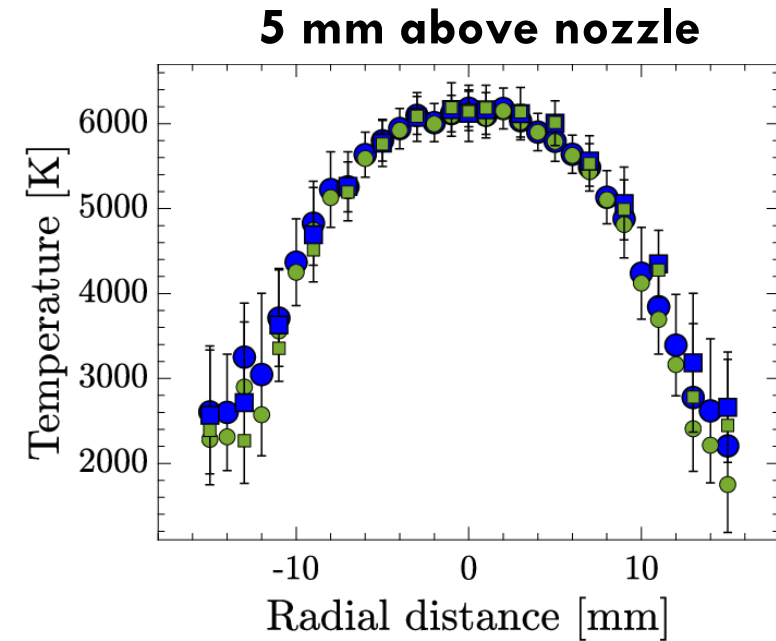
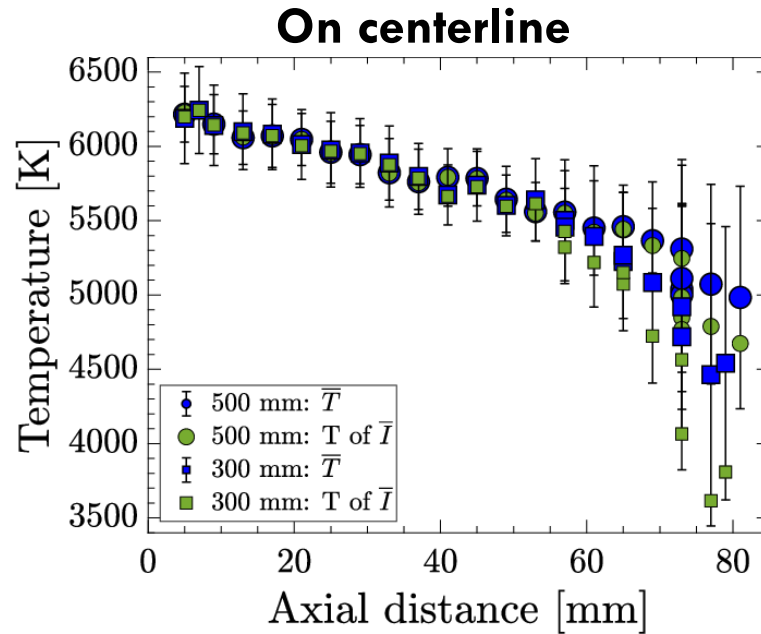
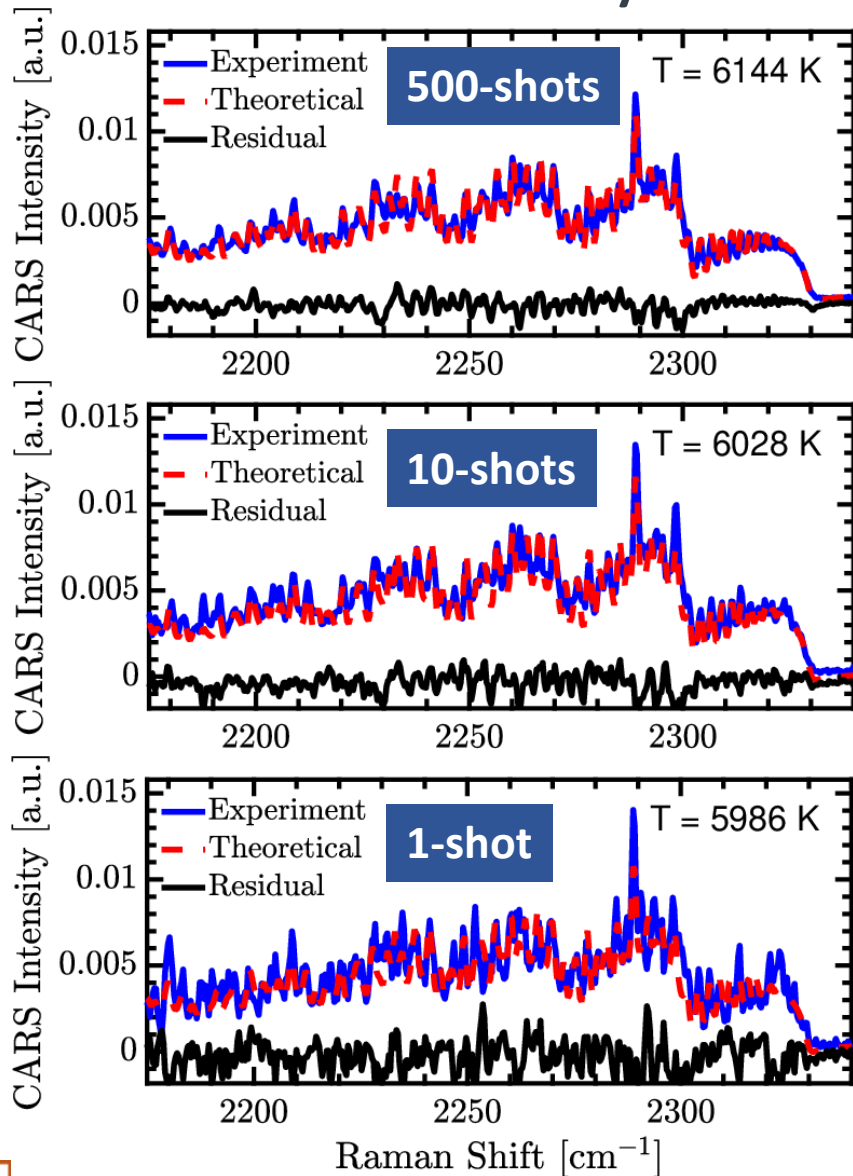
Broadband ns Multiplex CARS – N₂ Thermometry



0.75 m Czerny-Turner spectrograph, 1800 g/mm grating



N2 Thermometry



- Spectral fits with MATLAB implementation of Yuratich formalism [2]: temperature, non-resonant background.
- Temperature profiles: single shot and accumulated signal data.

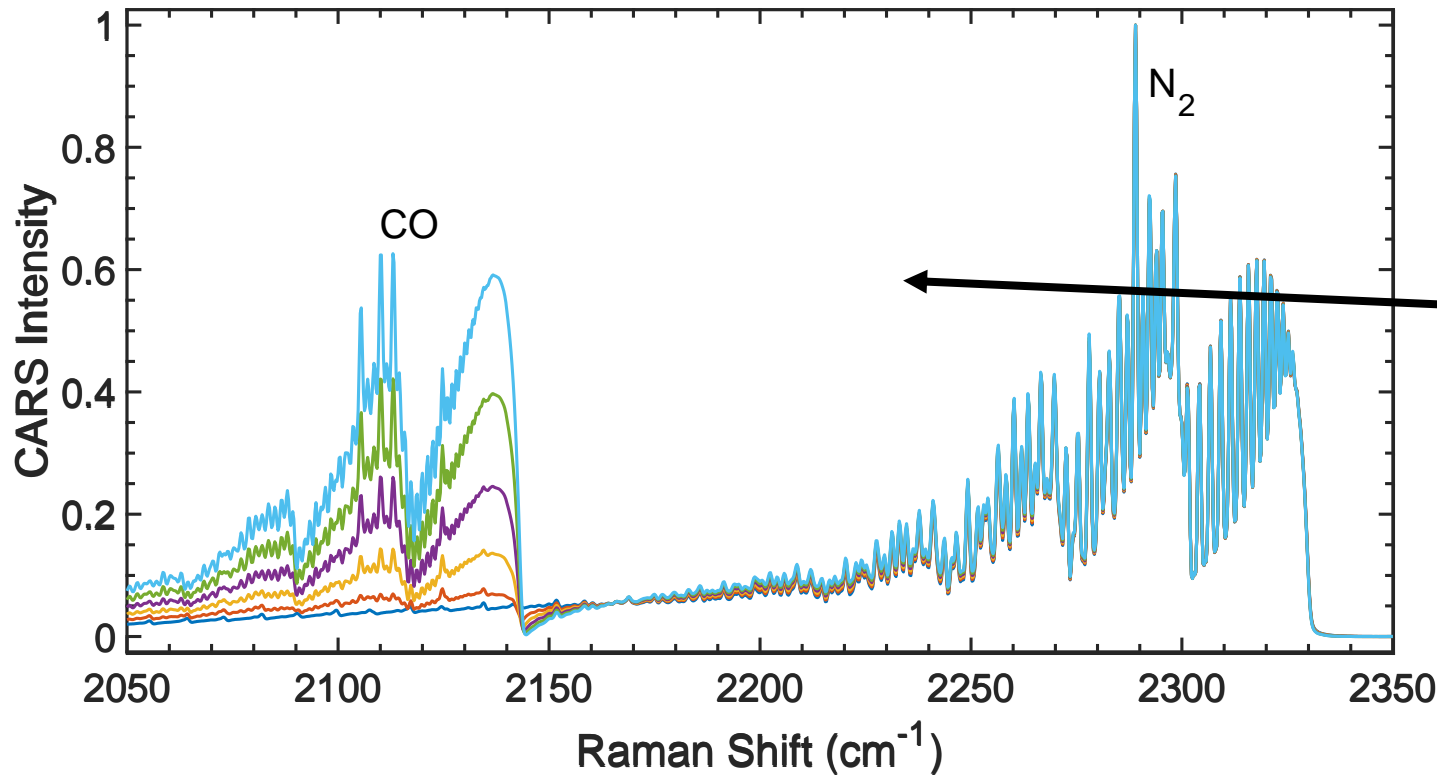
[1] Fries et al., "Nitrogen Thermometry in an Inductively Coupled Plasma Torch using Broadband Nanosecond Coherent Anti-Stokes Raman Scattering". *In preparation* (2023)

[2] Yuratich, "Effects of laser linewidth on coherent anti-Stokes Raman spectroscopy". *Molecular Physics* (1979)

Stokes Source Tuning and Optimization for N₂/CO

To probe N₂ and CO simultaneously:

- Need broad Stokes source spectral profile.
- Need sufficient sensitivity to relatively lower CO concentration.



Pump CO 10-20X harder!

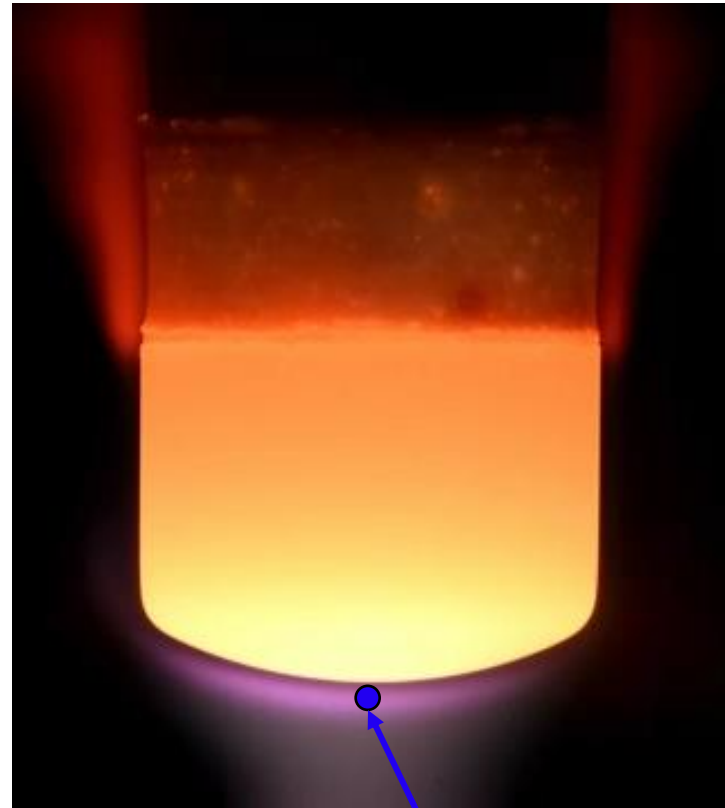
→ Rhodamine 610 and 640 to achieve $\lambda_c = 600 - 610$ nm and FWHM of 160-200 cm⁻¹.

Testing of Iso-q Graphite Samples

- Iso-q graphite sample with $\varnothing 30$ mm.
- CARS positional accuracy ± 80 μm
- About 90 s to steady state surface temperature and recession rate.
→ ~ 1 mm/min

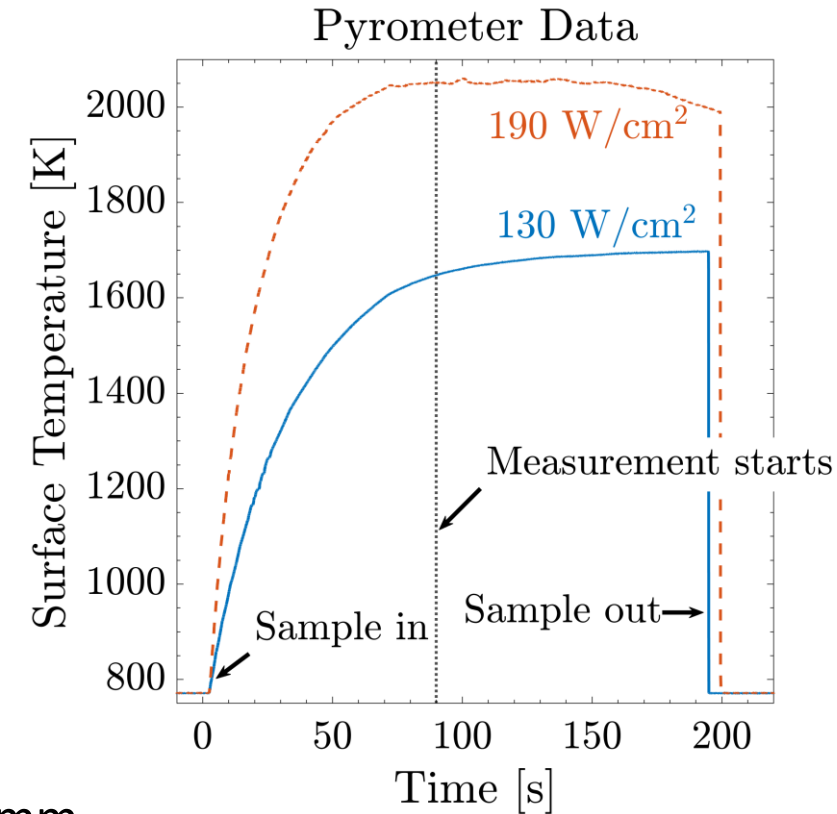
Two test cases

1. 10 kV anode voltage, 0.6 g/s tangential
→ ~ 130 W/cm^2 , $T_s \sim 1700\text{K}$ (avg.)
2. 11.3 kV anode voltage, 0.4 g/s tangential + 0.3 g/s axial
→ ~ 190 W/cm^2 , $T_s \sim 1980$ K (avg.)



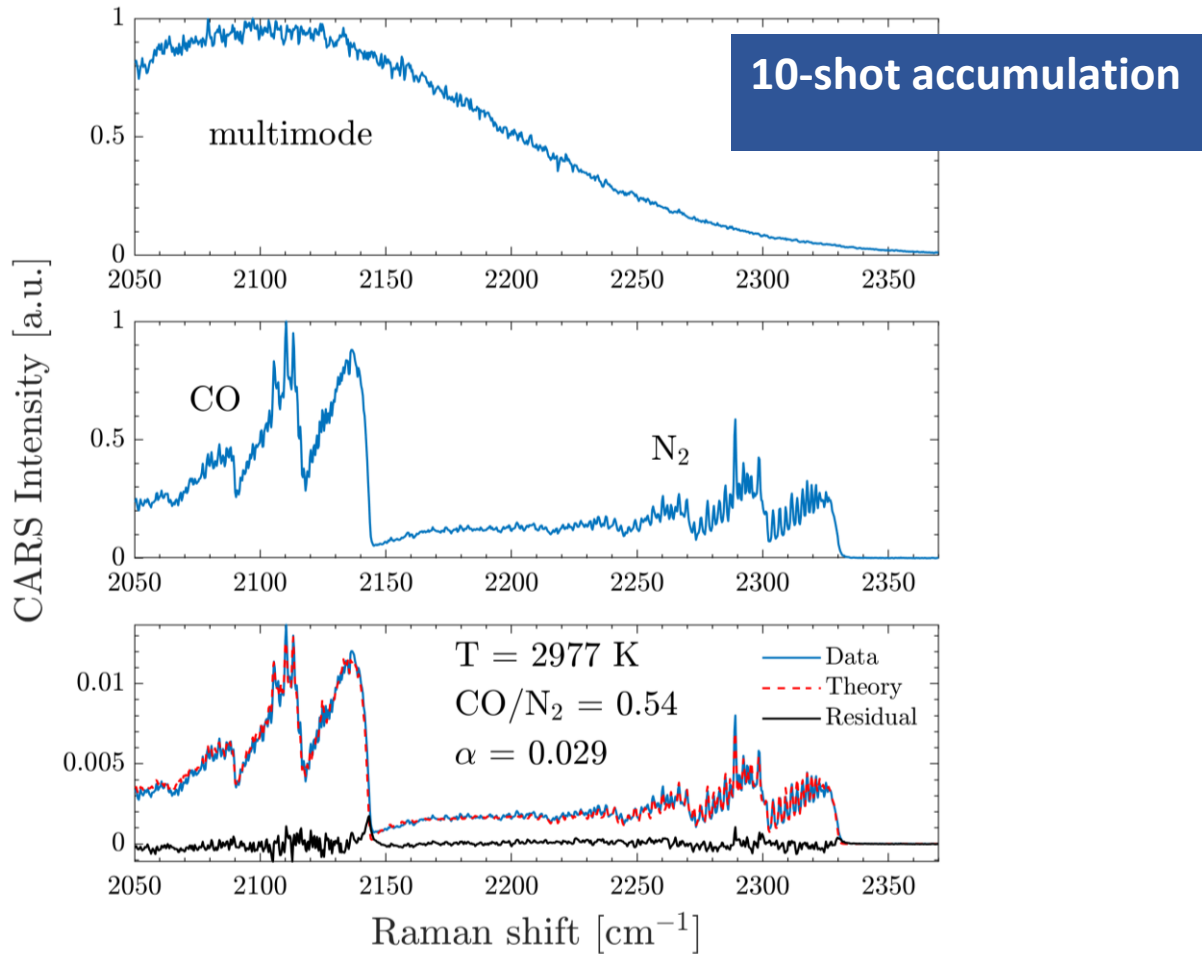
$T = 6000$ K
approach air stream

50 $\mu\text{m} \times 4.5$ mm
CARS volume

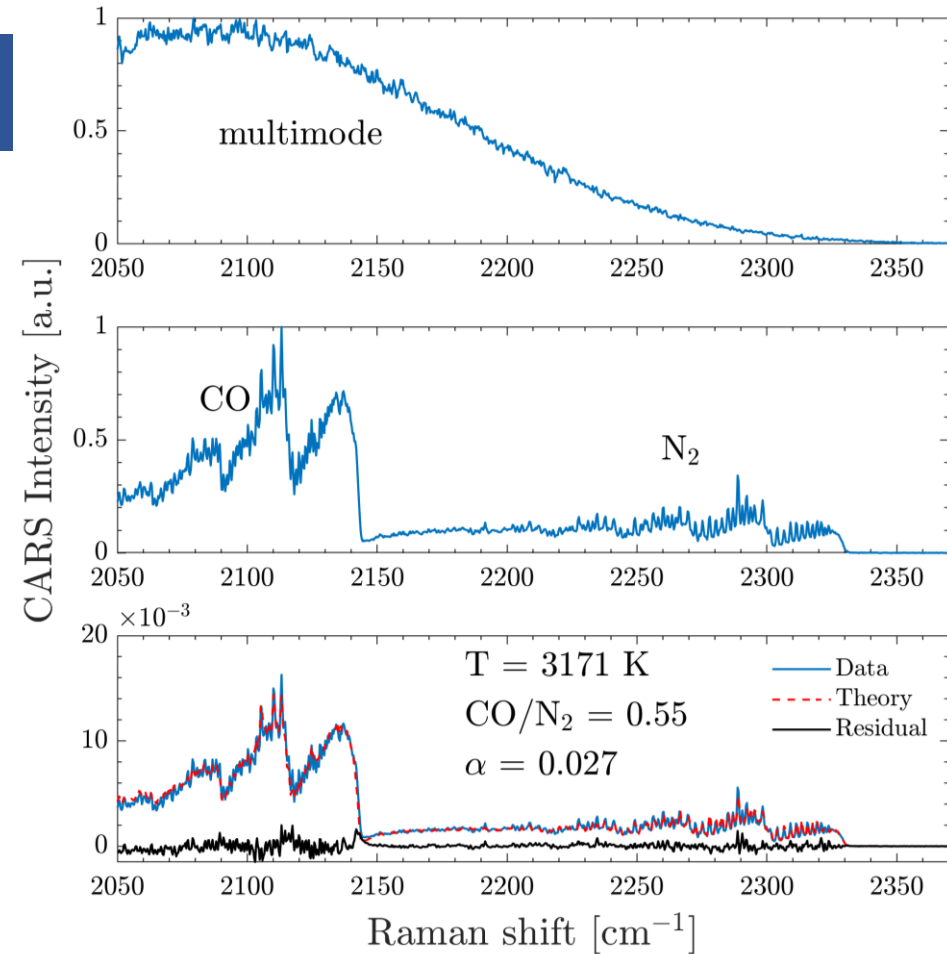


Broadband ns CARS – Simultaneous Probing of CO and N₂

Heat flux $\sim 130 \text{ W/cm}^2$

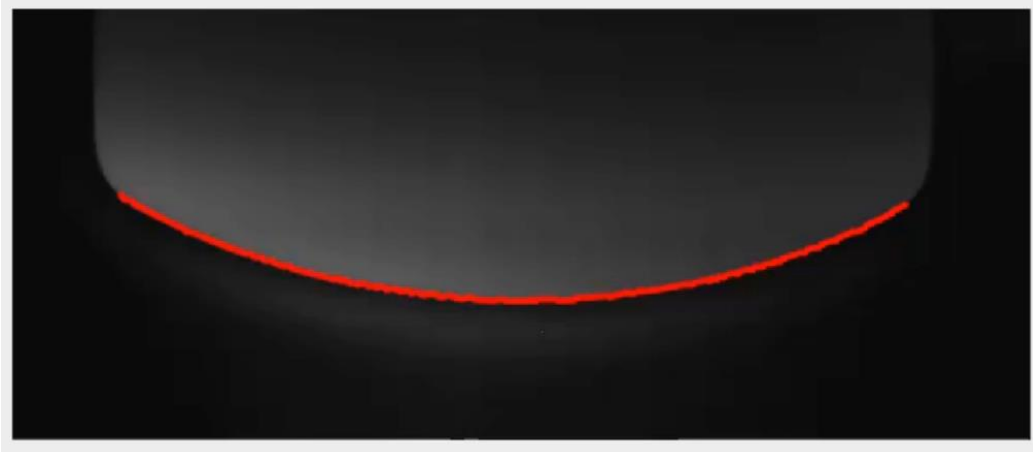


Heat flux $\sim 190 \text{ W/cm}^2$

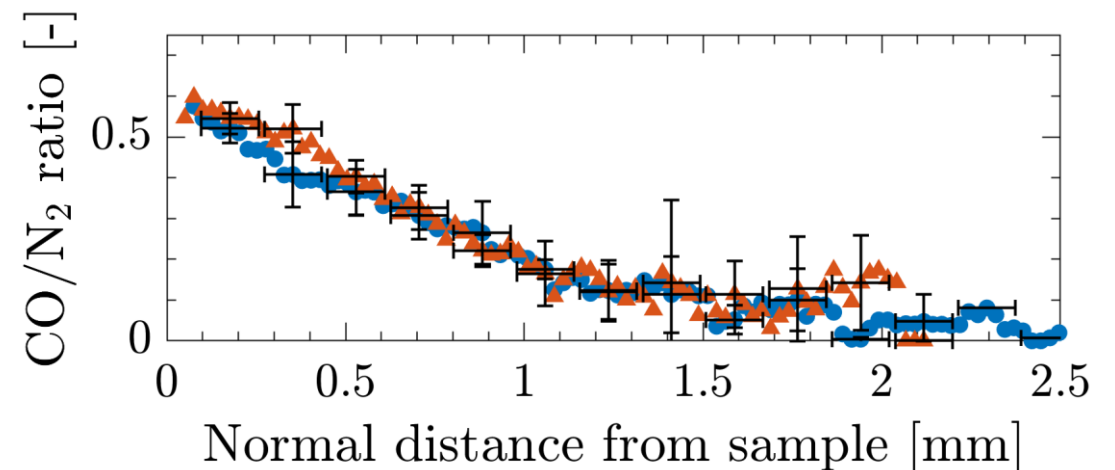
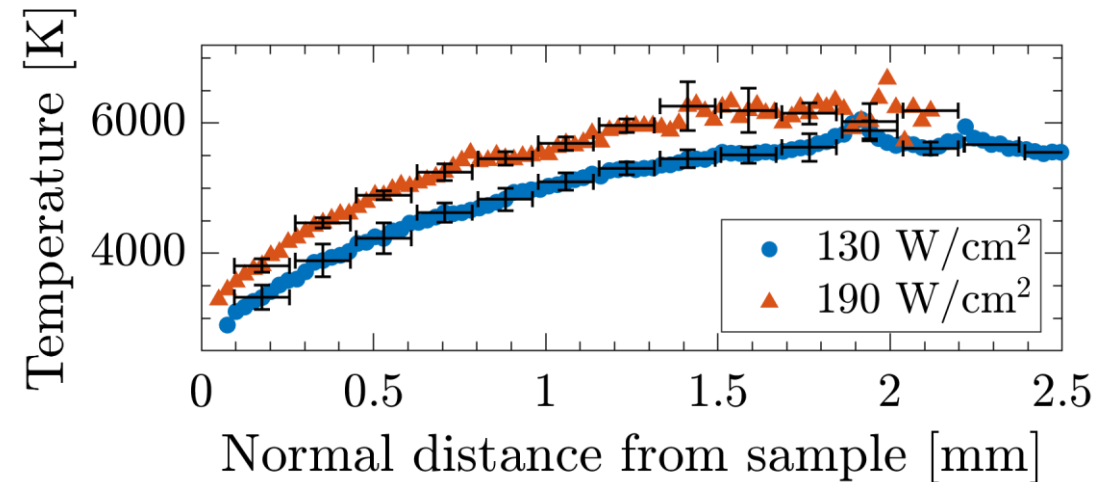
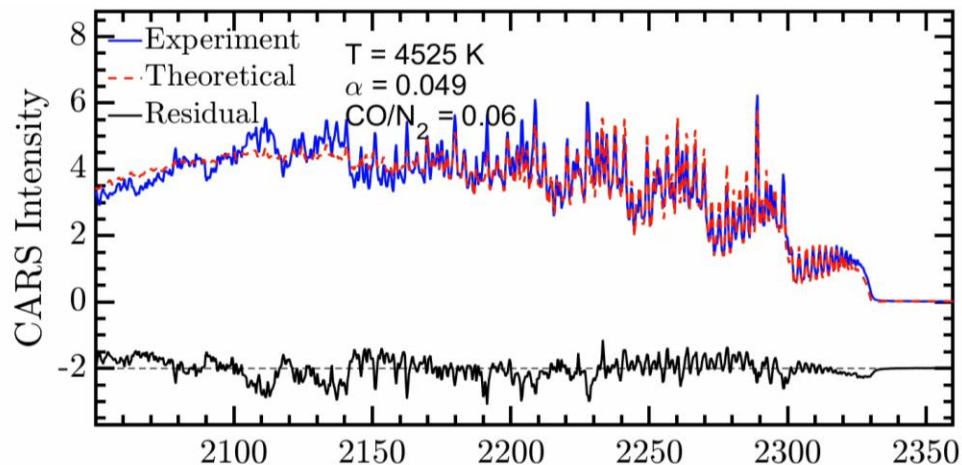


Temperature and Relative Concentration Profiles

Surface tracking



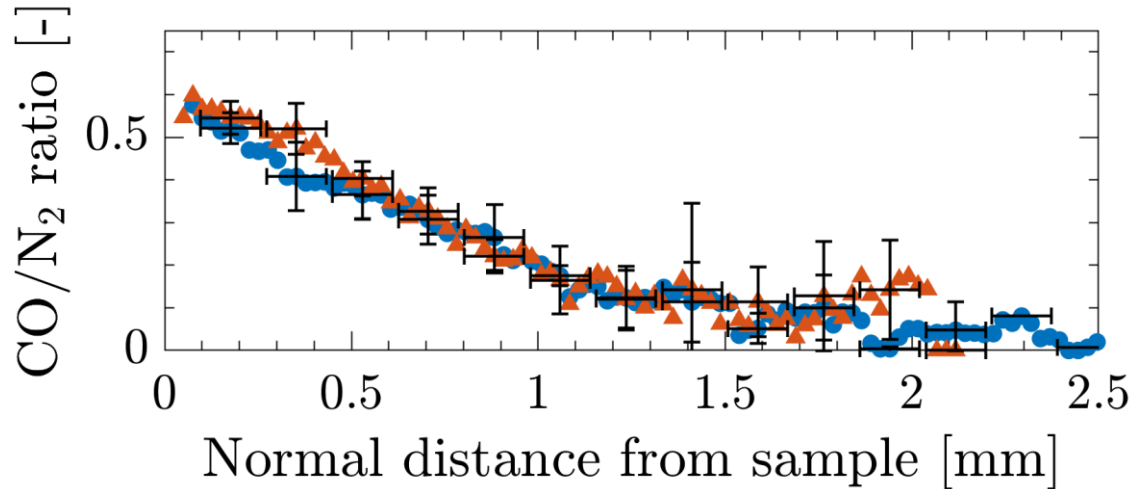
Spectrum evaluation



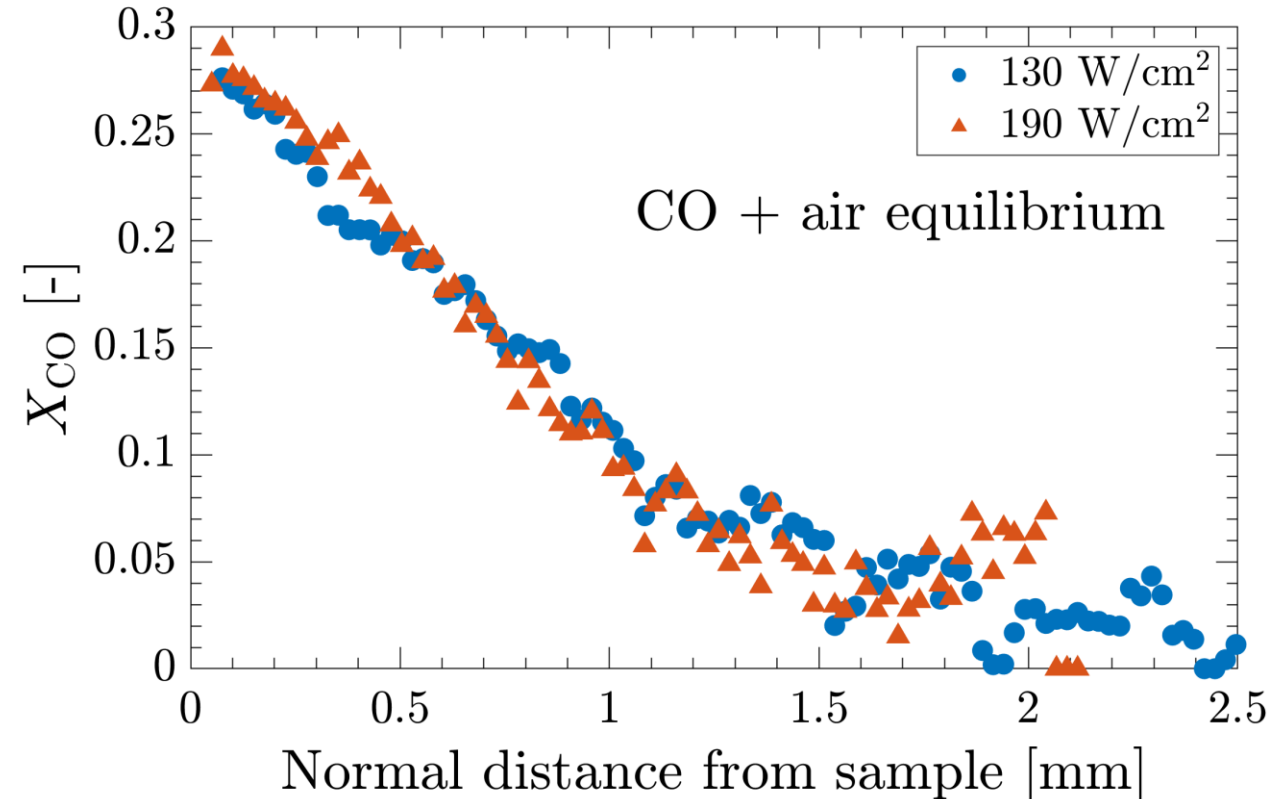
- Average of four data sets each.
- Vertical error bars represent 1σ of data series.

Relative Concentrations and Equilibrium Estimate

Relative CO mole fraction



Equilibrium CO mole fraction



CO mole fraction determined using NASA CEA equilibrium calculations:

- CO + air.
- Get X_{CO} via root finding at measured CO/N₂ ratio and temperature.

Conclusions

- Tailoring the Stokes spectral profile improves CO detectability.
- Peak CO concentration close to the sample surface:
 - ~60% of the N₂ concentration .
 - mole fraction of ~28% based on equilibrium CO + air mixture.
- Similar amounts of CO close to the sample surface for both \dot{q}'' cases.

Future Work

- Quantify detectability limit of CO, accuracy, and sensitivity of results to three-parameter fits.
- Compare measurements to phenomenological and finite rate graphite ablation models.

Thank you for your attention

Questions?

Poster 45

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